

AOE 4144: Applied CFD

A series of 12 lectures by Prof. Raj (course co-instructor)

Reflections on the Effectiveness of Applied Computational Aerodynamics for Aircraft Design

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<u>Lecture 12</u>

Topíc 6: ACA Effectíveness - Status and Prospects (3 of 3)Topíc 7: Closíng Remarks



List of Topics

Preface

- 1. Introduction
- 2. Genesis of Fluid Dynamics (Antiquity to 1750)
- 3. Fluid Dynamics as a Mathematical Science (1750–1900)
- 4. Emergence of Computational Fluid Dynamics (1900–1950)
- 5. Evolution of Applied Computational Aerodynamics (1950–2000)
 - 5.1 Infancy through Adolescence (1950–1980)

Level I: Linear Potential Methods (LPMs)

Level II: Nonlinear Potential Methods (NPMs)

5.2 Pursuit of Effectiveness (1980–2000)

Level III: Euler Methods

Level IV: Reynolds-Averaged Navier-Stokes (RANS) Methods

- 6. ACA Effectiveness: Status and Prospects (2000–20xx)
 - 6.1 Assessment of Effectiveness (2000–2025)
 - 6.2 Prospects for Fully Effective ACA (Beyond 2025)

7. Closing Remarks

Appendix A. An Approach for ACA Effectiveness Assessment



DNS–Seemingly The Only Option for Fully Effective ACA

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DNS can produce credible solutions <u>BUT</u> it will require incredible reductions in turnaround time and total cost for DNS to be fully effective in meeting aircraft design needs.

Both "A - Acceptance" and "Q - Quality" factors in $E = Q \times A$ need to be simultaneously maximized for Fully Effective ACA based on DNS

Since DNS is not expected to be feasible—even for a wing—until around 2080, <u>how could we improve</u> <u>ACA effectiveness in the interim</u>?



NASA CFD Vision 2030

Motivation "...the last decade has seen stagnation in the capabilities used in aerodynamic simulation within the aerospace industry, with RANS methods having become the high fidelity method of choice..."

> "...the <u>well-known</u> <u>limitations of RANS</u>

methods for separated flows

have confined reliable use

of CFD to a small region of

the flight envelope ... "

A Clarion Call to the Community in 2014

NASA/CR-2014-218178



CFD Vision 2030 Study: A Path to Revolutionary Computational Aerosciences

Jeffrey Slotnick and Abdollah Khodadoust Boeing Research & Technology, Huntington Beach, California

Juan Alonso Stanford University, Stanford, California

David Darmofal Massachusetts Institute of Technology, Cambridge, Massachusetts

William Gropp National Center for Supercomputing Applications, Urbana, Illinois

Elizabeth Lurie Pratt & Whitney, United Technologies Corporation, East Hartford, Connecticut

Dimitri Mavriplis University of Wyoming, Laramie, Wyoming



LES for Increased ACA Effectiveness

Pace of progress closely tied to advances in many key areas

- Grids: Methods for rapidly generating very fine, *truly* boundary-conforming grids
- Models: Advanced near-wall sub-grid-scale (SGS) models for WM-LES
- Algorithms: Higher-order numerical methods that minimize numerical dissipation
- **Software:** Development and implementation of effective strategies for designing computer software that exploits *emerging computer hardware architectures*
- V&V and UQ: Effective approaches for verification and validation of complex software, and for uncertainty quantification

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• Data Management: Costeffective approaches for efficiently managing large amounts of data, and for fast processing of extremely large datasets to extract information of value for ACA engineers **Computer Requirements**



• Etc., etc.

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ExaFLOPS Peak Speed is Within Reach L12



BUT We Need Sustained Speeds at ExaFLOPS Level for Practical LES Applications

Source: https://en.wikipedia.org/wiki/Supercomputer



State of the Art of LES and DNS (ca 2023)

Large Eddy Simulation (LES) of Turbomachinery Flow

Y. Fu, W. Shen, J. Cui, Y. Zheng, et al, Towards Exascale Computation for Turbomachinery Flows, SC'23, November 12-17, 2023, Denver, CO

Unstructured grid: 1.69 x 10⁹ elements High order solver: up to 8th order HPC cores: 19.2 million HPC performance: 115.8 PetaFLOPS (DP) *Gordon Bell Prize nominee*

Direct Numerical Simulation (DNS) of Full Aircraft Unstructured grid: > 10²⁰ elements HPC cores: ??? HPC performance: ???



<u>Evolutionary Path For</u> <u>DNS Faces</u> <u>Many Challenges</u>

How Do We Get There ("Nirvana") From Here?



Quantum Computing (QC) Offers A Ray of Hope!

Ongoing Revolutionary Research

- We can perform 2^N computations simultaneously on a quantum computer of N qubits (*qubits are quantum entities manipulated to act like computer bits*)
- A grid of 2^N elements requires a quantum computer with N qubits

Demo Problem	CFD	Demonstration	HPC cores	QC qubits
Turbomachinery	LES	2023 (now)	19.2 million	30
Full Aircraft	DNS	??? (future)	???	57

- Quantum-inspired, Hybrid Quantum-Classical, and Quantum Algorithms
 - An order of magnitude faster and cost-effective simulations using quantum algorithms than classical algorithms on today's HPC have been demonstrated
 - Quantum algorithms running on simulation platform of HPCs and quantum computers could reduce time and cost by nearly three orders of magnitude!

Potential for DNS of Full Aircraft Using QC Much Sooner Than The 2080s!

Source: Personal Communication, Abhishek Chopra, Founder & CEO, BQP (BosonQ Psi) (https://www.bqpsim.com)



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Conundrum for Today's Engineers

RANS is Here to Stay!

"...engineering calculations will have to be done by Reynolds-averaged methods for the foreseeable future..."

"...computer simulations of eddy motion can and will provide the detailed statistics—above all, the pressure fluctuation statistics—that cannot be adequately measured."

"...we cannot calculate all flows of engineering interest to engineering accuracy. However, the best modern methods allow almost all flows to be calculated to higher accuracy than the best-informed guess, which means that the methods are genuinely useful even if they cannot replace experiments."

> Excerpts from TURBULENT SECONDARY FLOWS by Peter Bradshaw Annual Review of Fluid Mechanics, Vol 19, 1987, pp 53-74

> > Author's Take: "Glass is half full"

Despite relatively low effectiveness for simulating complex flows, <u>RANS methods can, and do, add value if used wisely</u>



The TiCTaC Paradigm for Improving RANS-based ACA Effectiveness

<u>Tightly Coupled Tests and Computations</u>

A Near Term "Stopgap" Strategy

Devise the best way of *judiciously* coupling wind tunnel testing (WTT) with RANS CFD to deliver <u>credible</u> aerodynamic solutions—rapidly and affordably

2002: First proposed (Raj) NASA/DOD Workshop on Aerodynamic Flight Predictions, Williamsburg, VA, USA, Nov 19-21, 2002

Aerodynamic Flight Prediction: Apath Forward A 9. Goal: Rely on Computational Simulations for Aerodynamic Flight Prediction CFD and its Interdisciplinary Extensions 9. Problem: CFD Codes Will NOT Produce Credible Data for Your Application Unless Previously Validated on Same Application 1. Too Many Potential Traps: Computing Systems, Coding Errors, Complexity of Mathematical and Numerical Methods, Modeling Approximations, Inappropriate Use, etc., etc. 9. Approach: Develop and Implement "Validation Plan" Targeted at Maximizing Prediction Credibility for Your Application 1. Requires Dedicated Physical Experiments on Relevant Isolated Phenomena to Reduce and Quantify Uncertainties that Contribute to Lack of Confidence in Predictions

 Sandia National Lab's Accelerated Strategic Computing Initiative Provides a Good Starting Point

Adopt <u>Tightly Coupled Test and Computations</u> (TiCTaC) Concept

2012: Revisited (Raj)

5th Symposium on Integrating CFD and Experiments in Aerodynamics, JAXA, Tokyo, Japan, Oct 3-5, 2012

A Near-term Alternative: TiCTaC (<u>Tig</u>htly <u>C</u>oupled <u>T</u>est <u>a</u>nd <u>C</u>omputations)

- <u>Premise</u>: CFD Codes Will NOT Produce Credible Data for Your Application Unless Previously Validated on the "Same" Application
- <u>Approach:</u> Develop and Implement "Validation Plan" Targeted at Maximizing Prediction Credibility for Your Application
 - Identify the principal source(s) of uncertainty related to modeling of relevant flow physics and numerics
 - Perform dedicated tests for the sole purpose of "refining" modeling parameters
 - Utilize updated models to maximize credibility of CFD simulations

Can We Realize its Enormous Potential in Practice?

2014 & 2016: An updated approach (Raj et al) Applied Aero Conference, Bristol, UK

- Develop and implement *TiCTaC*: leverage complementary strengths of CFD and EFD by exploiting ongoing technological advances in both WTT and CFD
 - ✓ WTT (Additive Manufacturing, Rapid Prototype Testing, Measurement Techniques)
 - ✓ CFD (Grid Adaption, High Performance Computing, Uncertainty Quantification)



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Appendix A. An Approach for ACA Effectiveness Assessment



L12 **Pursuit of Fully Effective Applied Computational Aerodynamics (ACA) Started Long Ago...**

"Both for engineering and for many of the less exact sciences, such as biology, there is a demand for rapid methods, easy to be understood and applicable to unusual equations and irregular bodies. If they can be accurate, so much the better; but 1 per cent, would suffice for many purposes." – Richardson, 1910

"Prospective users...rarely interested in whether or not an accurate solution of an idealized problem can be obtained, but are concerned with how well the calculated flow agrees with the real flow." – Hess and Smith, 1967

"The effectiveness of computational aerodynamics depends not only on the accuracy of the codes but to a very large degree—perhaps more than is generally appreciated—on their robustness, ease and economy of use." – Miranda, 1982

...and Continues Today!



ACA Effectiveness

Present Status (~2025)

RANS-based ACA is Unlikely to be Fully Effective Anytime Soon, If Ever!

Future Prospects (beyond 2025)

Long-term: DNS-based ACA May Lead to Fully Effective Capability in the 2080 Time Frame Many Decades Hence—A Bridge Too Far?

Mid-term: LES-based ACA Offers a Promising Alternative in the 2050 Tiem Frame

Near-term: TiCTaC (Judicious Coupling of Wind Tunnel Testing and RANS CFD) Offers a "Stopgap" Option



Author's Advice for ACA Engineers

Use Effectiveness as a Guide to Wisely Choose CFD Codes for ACA Studies to Support Aircraft Design

1. Understand the Problem

 Develop a comprehensive understanding of the scope of customer needs (potential impact of solution, desired level of accuracy, type and amount of data, etc.) and constraints (cost and schedule)

2. <u>Devise an Approach</u> to Solving the Problem

- Examine <u>all four levels</u> of available CFD codes for solving the problem with effectiveness as the key measure of merit; "Know the Flow"
- Choose a code that can best meet customer's needs and constraints [consider the type, amount, and quality of the required aerodynamic data along with the specified constraints in choosing the code]

3. <u>Execute</u> to Generate and Deliver a *Best Solution* that Adds Value

Provide a solution that *best* meets customer needs while satisfying all constraints

Don't Use a Hammer When You Need a Screwdriver!



My Top 10 Takeaways From my journey on a long and winding road

1. ACA is an engineering discipline enabled by CFD

- CFD is to ACA as airplane is to air transportation.
- CFD produces data, à la EFD. ACA produces solutions.
 Don't confuse data with solution!
- ACA uses CFD to create <u>value</u> by delivering *credible solutions,* on time & on budget, to meet customer's pressing needs.



- 2. EFD remains the best source of data for assessing CFD 'goodness'
 - If CFD and EFD data don't match, ask why? If they do, most definitely ask why?

3. Effectiveness is the best Measure of Merit for Assessing ACA

- Effectiveness = quality x acceptance: E = Q x A
- ACA Effectiveness is ultimately assessed by design teams (who initiate the "Value Chain") in collaboration with ACA engineers—not by CFD code developers.
- 4. Predicting aerodynamic characteristics isn't that hard...making credible predictions is. And it's <u>REALLY HARD</u>!
- 5. Converting basic research concept into an *effective* capability is a long and arduous process marked by *invention, initiative, and innovation...and* lots of patience!*





My Top 10 Takeaways (contd.) From my journey on a long and winding road

6. Communication, Coordination & Collaboration across <u>all</u> stakeholders is essential to succeed in any endeavor.

7. Talent trumps tools, any day of the week!

- *It's the airplane, not the tools, stupid!*
- A talented engineer can do wonders even with poor tools. With proper tools, a talented engineer makes impossible possible!
- What matters most to the customers is results, not tools.
- Use Your Talents!
- 8. Nothing—*absolutely nothing*—is worth compromising your integrity.
- 9. Your ability to learn, not just what you know, is a key differentiator.
- 10. Life is akin to an unsteady system with unsteady boundary conditions, *don't expect a steady solution.*
 - Don't underestimate the role of luck!

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Be Mindful of

Four "Immutable" Laws and Principles!

Murphy's Law ''If anything can go wrong, it will.''

Parkinson's Law

"Work expands so as to fill the time available for its completion."

The Peter Principle

"In a hierarchy, every employee tends to rise to his level of incompetence."

The Dilbert Principle

"Companies tend to systematically promote their least-competent employees to management (generally middle management) in order to limit the amount of damage they are capable of doing."

You will never be disappointed in your professional life!



"Look ahead where the horizons are absolutely unlimited"



Robert E. Gross

Entrepreneur, Industrialist Founder, Lockheed Aircraft Corporation (now Lockheed Martin) Enshrinee, The National Aviation Hall of Fame 11 May 1897 – 3 Sep 1961

20



VITA

Pradeep Raj is currently a collegiate professor emeritus in the Kevin T. Crofton Department of Aerospace and Ocean Engineering at Virginia Tech in Blacksburg, Virginia, USA. Since 2012 through 2024, his primary responsibilities at VT included (a) teaching the AOE undergraduate Capstone Aircraft Design courses, and (b) pursuing collaborative research in multidisciplinary analysis, design, and optimization (MADO) tools and processes, with emphasis on applied aerodynamics, to enable simulation based design of affordable flight vehicles.

Raj joined VT in 2012 after a 32-year career with Lockheed Martin. Starting out in 1979 as a Senior Aerodynamics Engineer at the Lockheed-California Company in Burbank, he assumed positions of increasing responsibility before retiring in 2011 as a Director from Advanced Development Programs, Palmdale, California, commonly known as the Skunk Works[®] and widely recognized for creating breakthrough technologies and landmark aircraft. For the first 20 years, including 8 as Technical Fellow (1991-1999), Pradeep was instrumental in enhancing the effectiveness of applied computational aerodynamics for aircraft design through several research, development, and application campaigns.

Prior to joining LM, Raj spent one year (1978-79) as an assistant professor at the University of Missouri-Rolla (now Missouri Science & Technology University), and two years (1976-78) as a research assistant professor at the Iowa State University, Ames, Iowa. Pradeep earned his Ph.D. in Aerospace Engineering from Georgia Tech in 1976. Before going to GT, he earned a Master of Engineering (with Distinction) in Aeronautical Engineering in 1972, and a Bachelor of Engineering (with Distinction) in Electrical Technology in 1970, both from the Indian Institute of Science, Bangalore, India. He received a Bachelor of Science (with Honors) from Meerut University in 1967.

Pradeep Raj is a Fellow of the American Institute of Aeronautics and Astronautics (AIAA), of the Royal Aeronautical Society (RAeS), and of the Institute for the Advancement of Engineering (IAE).

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