



# *A Case Study of the Boeing B-47 Stratojet*

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# *Outline*

- u The political and economic climate
- u The evolution of the B-47
- u B-47 innovations in aerodynamics, stability and control, and structures
- u The legacy of the B-47



# *At This Time in History ...*

- u It's 1943 and World War Two is raging
- u Aircraft production in the U.S. is staggering
- u Aircraft development is rapid
  - Piston engine technology reaching limits
  - Turbojet engines show promise
  - Britain and Germany have advantage
  - Reports of German fighter jets prompt U.S. to consider jet bombers



# *Origins of the B-47*

- u **Air Corps Design Competition, 1944**
  - Design high speed jet bomber, 500 mph +
  - 8,000 pound payload
  - Range of 2,500 to 3,500 miles
  - Service Ceiling of 40,000 feet
  
- u **Designs submitted in 1944**
  - North American B-45
  - Convair B-46
  - Boeing B-47
  - Martin B-48



# *Evolution of the B-47*

- u B-47 began as straight wing design
  - Limited top speed to under 500 mph
- u Operation Paperclip
  - Adolph Busemann
  - Swept wing theory in 1935
- u Engines and more engines
  - Fuselage and wing mounted
  - Four or six?



# *Evolution of the B-47*

<b>Model #</b>	<b>Date</b>	<b>Wing</b>	<b>Engines</b>
413	Jan. 1944	Straight	4 Turbojets
422	early 1944	Straight	4 Turboprops
424,425	early 1944	Straight	4 Turbojets
426	early 1944	Straight	4 Turboprops
432	Dec. 1944	Straight	4 Jets in Fuse.
446	Sept. 1945	Swept	4 Jets in Fuse
448	Sept. 1945	Swept	6 Jets in Fuse.
448-2-2	Sept. 1945	Swept	6 Jets in Fuse.
450	Oct. 1945	Swept	6 Jets under wings



# *Evolution of the B-47*

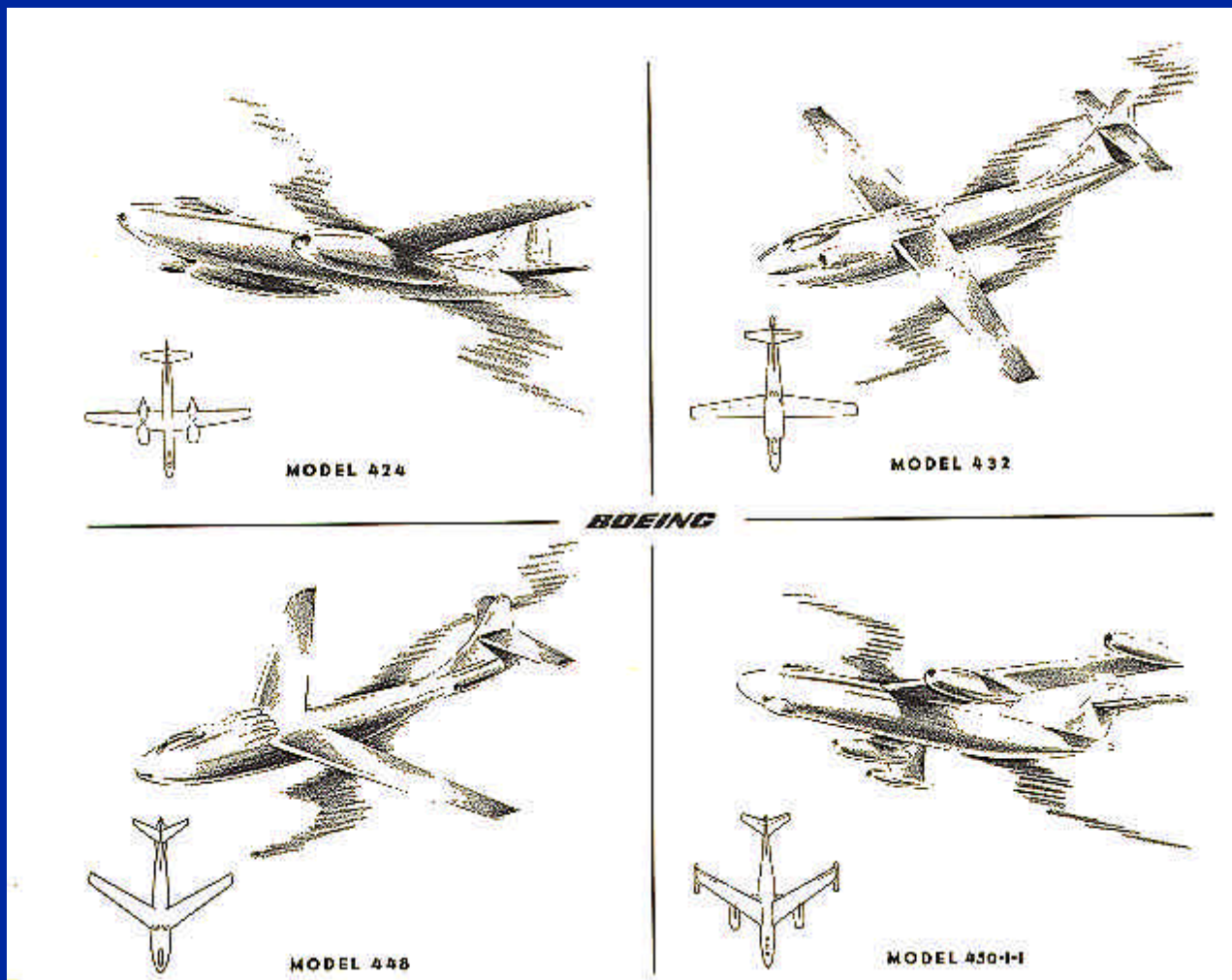
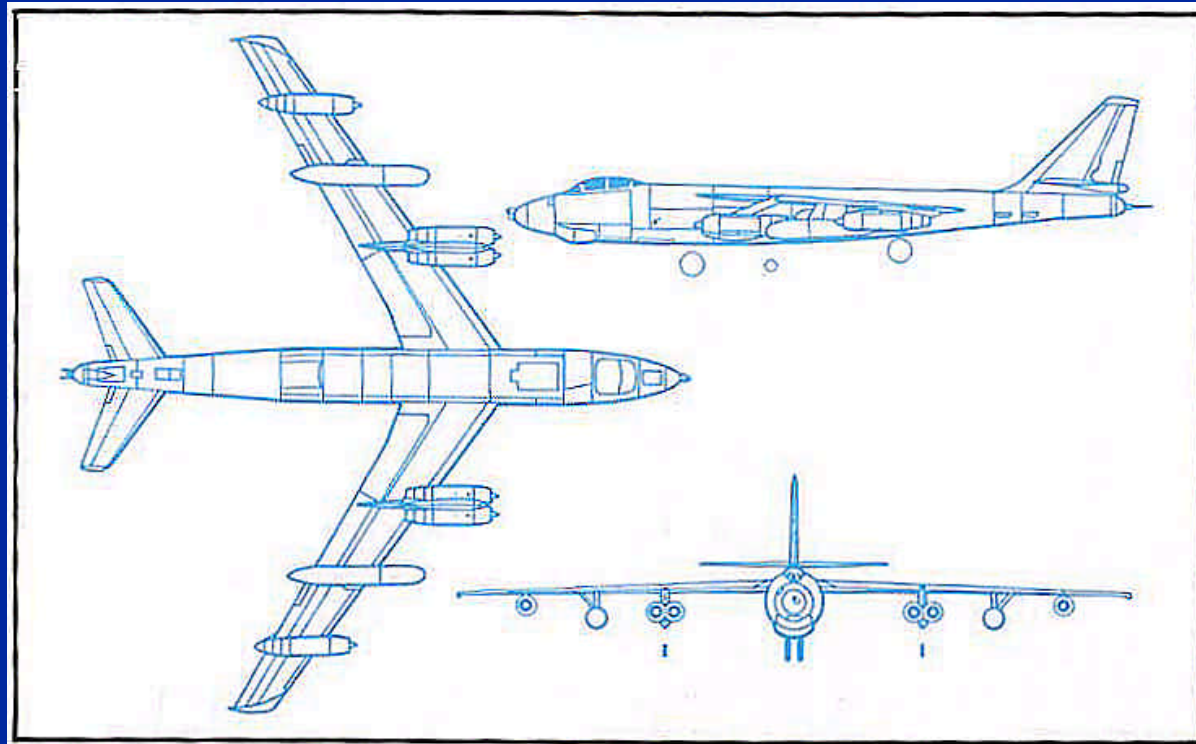


Image from: [The Politics of the U.S. Strategic Bomber Program](#)

Michael E. Brown, 1992.

# The Production B-47



Images From: [American Warplanes](#), Bill Gunston, 1986.



# *Podded Engines on the Wing*



Image From: [American Warplanes](#), Bill Gunston, 1986.



# *Podded Engine Advantages*

- u Interference drag was reduced in the wind tunnel
- u Strut weight was minimal
- u Increased span loading
  - Reduced the wing root bending moment
  - Reduced wing weight
- u Forward mass increases flutter resistance
- u Increased accessibility for maintenance
- u Increased passenger safety in case of an engine fire



# *Pitch-up Problems (Dana)*



# *Longitudinal Stability (Dana)*



# *The Yaw Damper (Dana)*



# *The Flexible Wing*

- u Long range transonic cruise requirements:
  - Large span for minimum induced drag (116 ft.)
  - Thin airfoils for minimum wave drag (12%)
- u This led to a very flexible wing
- u The load distribution was shifted inboard at high g's
  - This allowed an 8 ft. span extension at each tip without changing the original wing structure or engine locations
- u Acted as a shock absorber to reduce gust loads



# *The Legacy of the B-47*

- u The B-47 led directly to the development of the 707
- u Today's commercial transports follow the design paradigm of the B-47/707



Boeing 720  
1960



Boeing 777  
1996



# References

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- u Irving, Clive, *Wide-Body: The Triumph of the 747*, William Morrow, New York, 1993.